AP Disc Button.....

NOTE: The autopilot may be re-engaged. Refer to the normal procedures section of the AFMS, MANUAL PITCH TRIM WITH AUTOPILOT ENGAGED

WARNING: In flight, do not overpower the autopilot. The trim will operate in the direction opposing the overpower force, which will result in large out-of-trim forces. Do not attempt to re-engage the autopilot or use manual electric pitch trim until the cause of the malfunction has been corrected.

## **OVERSPEED PROTECTION**

"MAXSPD" & "AIRSPEED-AIRSPEED" airt

Aircraft Attitude and Altitude ... MONITOR After overspeed condition is corrected:

Autopilot ......RESELECT Throttle ...... AS NECESSARY

NOTE: Overspeed Protection Mode provides a pitch up command to decelerate the airplane at or below the maximum autopilot operating speed.

## **UNDERSPEED PROTECTION**

"MINSPD" & "AIRSPEED-AIRSPEED" airt

Throttle.....INCREASE AS REQUIRED Aircraft Attitude and Altitude ... MONITOR

After underspeed condition is corrected: Autopilot .....RESELECT

Throttle......REFERENCE NOTE: Underspeed Protection Mode provides a pitch down

command to maintain 65 KIAS.

## PITCH TRIM FAILURE

"PTRIM" Indicates pitch trim servo failure Control Wheel ......GRIP FIRMLY AP DISC Button...... PRESS and RELEASE Be prepared for high elev. control forces Elevator Trim ..... manually RE-TRIM

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### **EIS FAILURE**

Loss of displayed information on the EIS, incl. a blank, frozen, red 'X' over the I display, or unresponsive display

Engine controls...... ADJUST

Ensure operation within engine limitations

## **ESP ACTIVATION**

Throttle .....AS REQUIRED Aircraft Attitude.MAINTAIN/REGAIN CNTR

NOTE: If ESP is active for approximately 10 seconds, the autopilot will automatically engage in LVL mode, an aural 'ENGAGING AUTOPILOT' will be heard, and the autopilot will roll the wings level and fly at zero-vertical speed. ESP may be disabled by pressing and holding the AP DISC button. Releasing the button will allow ESP to

## **ATTITUDE ALIGNING**

"ALIGNING KEEP WINGS LEVEL" means that G3X detected invalid attitude input

Maintain 1° nose up pitch and wings level fliaht

NOTE: The system will display attitude when internal accuracy tolerances have been met.

Aircraft Attitude..LIMIT TO 10° bank, 5° pitch If attitude does not return, continue to use the standby flight instruments for aircraft attitude control

Legend: BOLD TEXT IN RED BOX = MEMORY ITEM

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# **ALERTS & ANNUNCIATIONS**

#### **WARNING ANNUNCIATIONS - RED ALERT ACTION** NOTE

Only climb maneuver is **TERRAIN** ecommended unless in OBSTACLE VMC or it is determined Disconnect AP and initiate that turning and climbing Terrain Ahead! Pull Up!" maximum performance is the safest course of Terrain, Terrain Pull Up! Pull climb (max T/O power and action. "Obstacle Ahead! Pull Up! Vx climb IAS) The arrow indicates the Obstacle, Obstacle Pull Up! errain is outside the Pull Un!

"Sink Rate, Pull Up!" A red X through any Reference the data source display field indicates that Red X or alternate equip is not receiving data, or is corrupted.

Synthetic Vision field of

annun aren't displayed.

he arrow indicates the terrain

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Take appropriate action to The engine parameter has Red Engine correct condition causing exceeded the warning engine parameter Parameter hreshold exceedance Manually fly the airplane. Autopilot has failed or is Silence the AP discn tone noperative and remove the annun, by pressing the AP annun in

the FD mode bar light Director and **AFCS** Manually fly the airplane utopilot have failed utopilot Failure Manually fly the airplane

Manually trim the airplane Electric pitch trim is **PTRIM** using the pitch trim wheel inoperative Select full screen mode on Display is in split screen display to view WARNING mode and WARNING

annunciations

ACTION **ALERT** 

TERRAIN Take corrective action until the is outside the Synthetic Visio **OBSTACLE** alert ceases. Using all available ← OBSTACLE information to determine the "CAUTION, Terrain" appropriate action, alter the Caution Terrain Ahead' flight path away from the threat "CAUTION, Obstacle" by stopping descent, climbing CAUTION, Obstacle Ahea and/or turning "CAUTION, Sink Rate" Manually fly the airplane Pilot has disconnected the AP. ne Electrical Control System ECS FAIL s failed (GAD 27 FAILED) direction The autopilot is holding ↑TRIM UP1 nose up Move the nose aircraft being of of trim due to annun ↓TRIM DOWN↓ changes in airspeed or power

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nose to reduce airspeed maximum airspeed limit Displayed heading is outside of Use standby compass (amber bcgr ne internal accuracy limits. Display is in split screen mode elect full screen mode on display to view **CAUTION** annunciations ot displayed Fly aircraft manually and crosscheck Attitude and Heading Reference attitude indication with standby attitude system is aligning. AHRS may no AHRS ALIGN indicator and other sources of attitude lign with excessive pitch/bank Keep Wings information. Limit aircraft attitude to ±10° Level bank and ±5° pitch as AHRS Aligns - OK to taxi

Add maximum available power. Autopilot Airspeed is too slow,

educe power. Autopilot will raise aircraft Airspeed is approachin

will lower aircraft nose to increase

airspeed

MAX SPEED

TAWS FAIL

ne AHRS monitors have Fly aircraft manually and crosscheck etected a possible AHRS attitude indication with standby attitude alfunction or an error with the AHRS ALIGN indicator and other sources of attitude ttitude presentation. The AHRS attempting to realign itself. he autopilot may automatically isconnect

Cross-check the flagged information ifference detected between t against other sources to identify G3X Touch and attitude display MISCOME erroneous information nd the G5 attitude display. Use standby attitude source, or, if AHRS he GSU 25 AHRS has failed AHRS FAI REVERT message is also displayed, ATTITUDE continue to use G3X Touch. Attitude will be from the G5

he GSLL25 AHRS has failed and Continue to use the G3X Touch ttitude from the G5 is being REVER' splayed on the G3X Touch Cross-check the flagged information ifference detected between the AIR MISCOM against other sources to identify G3X Touch airspeed or altitude IAS MISCOMI erroneous information nd the G5 airspeed or altitude. Use standby airspeed and altitude The GSU 25 air data computer indicato, or, if ADC REVERT message is as failed ADC FAIL

also displayed, continue to use G3X Touch he GSU 25 air data computer has failed and air data from the ADC REVERT Continue to use G3X Touch G5 is being displayed on the G3X

(Flashing) Press the flashing message annunciation new system message ha MESSAĞÉ to view a new system message nnunciated. Take appropriate action to correct he engine parameter has Amber FIS condition causing engine parameter ceeded the caution threshold exceedance De-select GPSS on the G3X Touch and he GPS Steering command to **GPSS** select desired alternate autopilot later he interfaced traffic system has Visually acquire the traffic to see and TRAFFIC letermined that nearby traffic avoid ay be a threat to the aircraft TAWS N/A.

ADVISORY ANNUNCIATIONS – WHITE **ACTION** NOTE ALERT

Use vigilance terrain depiction and

TAWS alerting is no longer provided

The unit will not be able to Cross check the information between the perform the miscompare NO COMP G5 and the PFD to determine which unit is in error monitor function

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Database errors or lack of GPS